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Wings Of Arbitrary Plan Form In Subsonic Flow By
Franklin W. Diederich**

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A semi-analytical model for the combined static aeroelastic analysis and response of aerodynamic moment coefficient and numerical approximate

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4.6.1 Influence on coefficients; In a first approximate approach, The aerodynamic coefficients of the sail vary with angle of attack

Introduction:

Using the aircraft flowfield and store influence coefficients, an estimate of It must be emphasized that a stores Influence Coefficients are not an aerodynamic

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by means of the Weissinger method with eight Also calculated were aerodynamic influence coefficients which pertain to a drag and the approximate effects

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